

filter 42, and a positive displacement main fuel pump 44, finally entering the fuel controller 40. It should be noted that the main fuel pump 44 is driven by the engine 10 and thus has a pump speed proportional to the engine speed.

As discussed in the preceding section, the main fuel pump 44 develops a volumetric flow rate dependent upon the pump shaft speed and is therefore sized to provide a fuel flow at the pump outlet 46 in excess of the metered fuel flow rate. The fuel controller 40 accepts the fuel from the pump outlet 46 and divides the flow stream between a supply line 48 which is routed to the combustion section 50 of the gas turbine engine 10, and a bypass line 52. The fuel flow rate in the supply line 48 is the metered fuel flow rate as determined by the fuel controller 40 while the fuel flow in the bypass line 52 is equal to the excess main pump fuel delivery.

In this first embodiment of the present invention, the bypass line 52 includes two branches, a first branch 54 and a second branch 56 together providing a means for returning and distributing the bypass flow between two return locations 58, 60, respectively. The first and second return locations 58, 60 are disposed respectively upstream of the first fuel-oil heat exchanger 20, and intermediate the first and second fuel-oil heat exchangers 20, 28. The flow of bypass fuel is allocated between the locations 58, 60 by a diverter valve 62 operable between a first position wherein the entire flow of bypass fuel in the bypass line 52 is directed to the first return location 58, and a second position (not shown) wherein the entire bypass fuel flow is directed to the second location 60. It should be noted at this time that although the diverter valve 62 is disclosed as operating in an either/or fashion for diverting the entire bypass fuel stream, it may be useful under some circumstances to employ a partial diverter valve operable for dividing the bypass fuel between the first and second branches 54, 56 in a proportional manner.

It is preferable to operate the diverter valve 62 responsive to an engine operating parameter related to the rate of heat rejection to the oil loops 14, 16. One such parameter is the fuel pressure rise across the engine driven boost pump 36 which is related to engine speed.

In operation, fuel and oil flow in the above-described systems with heat exchange therebetween accomplished in the fuel-oil heat exchangers 20, 28. Under conditions of low engine power, such as idling either on the ground or in flight, the metered fuel flow rate is relatively low, matching the fuel demand of the engine 10. As the engine shaft speed at idle is also relatively low as compared to cruise or full power levels, the output of the positive displacement main fuel pump 44, although much greater than the metered fuel flow rate, is also reduced. The diverter valve 62 is positioned during these periods to direct the entire bypass fuel flow to the first return location 58 through the first return branch 54. In this configuration, the entire bypass fuel flow and metered fuel flow pass sequentially through the first and second fuel-oil heat exchangers 20, 28.

During extended periods of idling resulting in excessive heat buildup in the recirculating fuel, the first fuel-oil heat exchanger 20 acts to remove heat from the fuel by transferring heat in the reverse direction into the first oil loop 14. This heat is removed from the loop 14 by opening the valve 24 to admit a flow of cooling air 22 through the first air-oil cooler 18. Similarly, during periods of inflight engine shutdown, heat removed from the windmilling engine, accessory drive, and recirculat-

ing fuel is rejected from the system through the air-oil coolers 18, 26.

During periods of full power or cruise engine operation, the diverter valve 62 is moved to the second position wherein the entire flow of bypass fuel is returned to the second return location 60 through the branch 56. In this configuration, the fresh supply of fuel from the fuel tank 34 forms the entire fuel flow through the first fuel-oil heat exchanger 20 wherein the fuel absorbs heat from the circulating oil in the first loop 14. The second fuel-oil heat exchanger 28 receives both the bypass fuel returned by the controller 40 as well as the fuel flowing from the first fuel-oil heat exchanger 20. This combined fuel flow passes through the second fuel-oil heat exchanger 28, cooling the oil circulating in the second oil loop 16, and passing subsequently through the filter 42 and main fuel pump 44.

It will be appreciated that during operation at these higher power levels, both the metered fuel flow rate and the main fuel pump delivery rate are considerably higher than those under idle conditions. The high metered fuel flow rate provides adequate total heat capacity in the supplied fuel stream for absorbing all the heat energy generated by the accessory drive 12 and the engine 10 thus allowing closure of the first and second airflow regulating valves 24, 32 improving overall engine efficiency.

Additionally, by redirecting the bypass fuel return flow from the first location 58 to the second location 60 downstream of the first fuel-oil heat exchanger 20 increases the temperature effectiveness of the first fuel-oil heat exchanger 20 which receives only fresh fuel from the fuel tank 34, unmixed with the warmer bypass fuel stream. This flow configuration insures that the maximum cooling capacity of the fresh fuel stream is available to the accessory drive unit 12 through the first oil cooling loop 14 when the engine operates at full or cruising power.

One final feature of the embodiment of FIG. 1 are oil bypass lines 64, 65 disposed in the oil loops 14, 16 for directing oil around the respective fuel-oil heat exchangers 20, 28. The bypass flows are regulated by control valves 66, 67 which are opened responsive to fuel and oil temperature during periods, such as at idle, wherein the fuel is too hot to absorb additional heat energy, thereby allowing the system to more flexibly accommodate the needs of the various systems.

By placing the fuel-oil heat exchangers 20, 28 upstream of the main fuel pump 44 and the fuel filter 42, the heat management system according to the present invention also reduces or eliminates the need for auxiliary fuel heating to avoid icing up of the fuel filter 42 under extremely cold operating conditions.

FIG. 2 shows a schematic representation of a second embodiment of the heat management system according to the present invention wherein like reference numerals are used to denote elements in common with the embodiment shown in FIG. 1. The second embodiment according to the present invention distributes the bypass fuel flowing in bypass line 52 between two return locations on the low pressure side of the main fuel pump 44, a first location 58 via a first branch 54, and a third location 68, via a third branch 70. It will be appreciated that the return location and branch denoted by reference numerals 68 and 70, while forming the only other location and branch in the disclosed second embodiment according to the present invention, are termed the third location and third branch to distinguish from the